

775

PIONEER 10
72-012A-00H

PIONEER 11
73-019A-00G

TRAJECTORY DATA - SFDU FORMAT

REQ. AGENT

CMW

ACQ. AGENT

JFC

PIONEER 10

72-012A-00H

TRAJECTORY DATA - SFDU FORMAT

This data set consists of 2 magnetic tapes. The tapes were written on 9-track, 6250 bpi, in SFDU format, and can be read with the VAX COPY command in VMS directory format. The first two files on the tape contains the volume description and the file format. These are followed by the data files. All data records are of the same length, 512 ASCII characters or bytes per physical record. The C tapes are 3480 bpi. The D and C numbers and time span are as follows:

D#	C#	FILES	TIME SPAN
-----	-----	-----	-----
D-085883	C-029141	39	04/06/73 - 01/01/1990
D-108359	C-032396	23	01/01/90 - 01/01/2000

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D#	C#	FILES	TIME SPAN
-----	-----	-----	-----
D-085884	C-029142	37	04/06/73 - 01/01/1990
D-108360	C-032397	23	01/01/90 - 01/01/2000

CCSD3ZF0000100000001CCSD3VS00002MRK**001

VOL_IDENT: USA_NASA_NSSD_P10Z_0002
VOL_CREATION_DATE: 1995-07-11
MEDIUM_DESCRIPTION: 1/2-inch, 9-track, 6250 bpi magnetic tape
TECHNICAL_CONTACT: David W. Lozier
NASA Ames Research Center (Mail Stop 244-14)
Moffett Field, CA 94035

Electronic Mail: David_Lozier@QMGATE.ARC.NASA.GOV
Telephone: 415-604-3147

PREV_VOLS: USA_NASA_NSSD_P10Z_0001

CCSD\$\$MARKERMRK**001CCSD3SS00002MRK**002

DATA_SET_NAME: PIONEER 10 TRAJECTORY EPHEMERIS ARCHIVE (Extension to 2000)

DATA_SOURCE: Pioneer 10 trajectory data were generated using the Jet Propulsion Laboratory's (JPL) "Double Precision Trajectory" (DPTRAJ) computer program, an element of the JPL Navigation Software System, reference (A).

DPTRAJ creates a spacecraft ephemeris by integrating the equations of motion for a full set of acceleration models. For the Pioneer 10 trajectory ephemeris, DPTRAJ integration was limited to the N-body gravitational terms (planets, Sun and Moon) with Earth oblateness, Jupiter oblateness (J2 and J4), and a simple solar pressure model included.

Initial conditions for the DPTRAJ runs were obtained from various Orbit Determination (OD) solutions performed during the mission. A set of seven spacecraft initial state vectors (position, velocity and epoch) was compiled to cover all phases of the mission from launch to 1 January 1990 (Volume USA_NASA_NSSD_P10Z_0001).

- (1) Launch to second midcourse maneuver
- (2) Second midcourse maneuver to first "IO" targeting maneuver
- (3) First "IO" targeting maneuver to second "IO" targeting maneuver
- (4) Second "IO" targeting maneuver to Jupiter phase
- (5) Jupiter encounter OD
- (6) Jupiter encounter to 1983 solar escape
- (7) 1983 solar escape to 1 January 1990

A complete set of input for the seven DPTRAJ runs is provided in reference (B).

This volume extends the Pioneer 10 trajectory from 1 January 1990 to 1 January 2000.

The DPTRAJ input for volume USA_NASA_NSSD_P10Z_0002 is provided in reference (D).

SCIENTIFIC_CONTACT: none

SPACECRAFT_CHARACTERISTICS: Pioneer 10 is a 550 pound spin stabilized spacecraft powered by four RTGs. Communications with the Earth is provided by a 9 ft. diameter parabolic high gain antenna located on the spin axis of the spacecraft. Periodic precession maneuvers are performed to maintain Earth pointing of the high gain antenna. The spacecraft

FATS070 CONTROL CARD TABLE SIZE IS 4096 BYTES

IC-030396
1/1/1990 - 1/1/2000

1-- ANALYZE BLP, LABELS=NO, PRILEN=2000
2-- PRINT LF=1, B=1-29
3-- PRINT LF=2, B=1-47
4-- PRINT LF=3, B=1-5
5-- PRINT LF=22, B=15-24
6-- PRINT LF=23, B=1-4

FATS071 TAPE BUFFER SIZE IS 65535 BYTES

CHARACTERISTICS OF THE TAPE TO BE ANALYZED
UNIT SERIAL DEN TRICH
5B5 CMS20 38000

FATAR DETAIL REPORT

1...5...10...15...20...25...30...35...40...45...50...55...60...65...70...75...80
(COLUMN GRID IS VALID ONLY FOR CHARACTER FORMATTED DATA)

BLOCK LENGTH MESSAGE/
NUMBER DISPL BLOCK TYPE
***** START FILE 1

1 495 PRINT REQUESTED
+00080 CCSD3ZF00001000000001CCSD3VS00002MRK**001 VOL IDENT: USA NASA NSSD
+00160 _P10Z_0002 VOL CREATION DATE: 1995-07-11 MEDIUM DESCRIPTION: 1/2-inch, 9-t
+00240 rack, 6250 bpi magnetic tape TECHNICAL CONTACT: David W. Lozier NASA Am
+00320 es Research Center (Mail Stop 244-14) Moffett Field, CA 94035
+00400 Electronic Mail: David.Lozier@QMGATE.ARC.NASA.GOV Telephone: 415-604-
+00480 3147 PREV_VOLS: USA_NASA_NSSD_P10Z_0001 CCSD\$MARKERMRK**001CCSD3S500
002MRK**002

2 460 PRINT REQUESTED
+00080 T DATA_SET_NAME: PIONEER 10 TRAJECTORY EPHEMERIS ARCHIVE (Extension to
+00160 generated using the 2000) DATA_SOURCE: Pioneer 10 trajectory data were
+00240) "Double Precision + Jet Propulsion Laboratory's (JPL
+00320 gram, an element of the JPL Navigation Software System, reference (A).
+00400 DPTRAJ creates a spacecraft ephemeris by integrating

3 454 PRINT REQUESTED
+00080 F the equations of motion for a full set of acceleration models.
+00160 For the Pioneer 10 trajectory ephemeris, DPTRAJ integration was limited t
+00240 o the N-body gravitational terms (planets, Sun and Moon) with Earth oblat
+00320 eness, Jupiter oblateness (J2 and J4), and a simple solar pressure
+00400 model included. Initial conditions for the DPTRAJ runs were obtained
from various Orbit Determination (OD) solutions

4 455 PRINT REQUESTED
+00080 G performed during the mission. A set of seven spacecraft initial
+00160 1 state vectors (position, velocity and epoch) was compiled to cover all p
+00240 hases of the mission from launch to 1 January 1990 (Volume (1) launch
+00320 USA_NASA_NSSD_P10Z_0001).
+00400 to second midcourse maneuver targeting maneuver (2) Second midcourse ma
neuver to first "10"

5 469 PRINT REQUESTED
+00080 N second "10" targeting maneuver (3) First "10" targeting maneuver to
+00160 targeting maneuver to Jupiter phase (4) Second "10"
+00240 (5) Jupiter encounter OD (6) Jupiter encounter to 1

FATS070 CONTROL CARD TABLE SIZE IS 4096 BYTES

FATAR CONTROL CARDS

1-- ANALYZE BLP, LABELS=NO, PRTLEN=2000
 2-- PRINT LF=1, B=1-31
 3-- PRINT LF=2, B=1-47
 4-- PRINT LF=3, B=1-5
 5-- PRINT LF=22, B=15-24
 6-- PRINT LF=23, B=1-4

FATS071 TAPE BUFFER SIZE IS 65535 BYTES

CHARACTERISTICS OF THE TAPE TO BE ANALYZED
 UNIT SERIAL DEN TRICH
 5B5 CMS21 38000

FATAR DETAIL REPORT

1...5...10...15...20...25...30...35...40...45...50...55...60...65...70...75...80
 (COLUMN GRID IS VALID ONLY FOR CHARACTER FORMATTED DATA)

BLOCK LENGTH MESSAGE/
 NUMBER DISPL BLOCK TYPE

1 495 PRINT REQUESTED
 +00080
 +00160
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 +00320
 +00400
 +00480

2 500 PRINT REQUESTED
 +00080
 +00160
 +00240
 +00320
 +00400
 +00480

3 466 PRINT REQUESTED
 +00080
 +00160
 +00240
 +00320
 +00400

4 461 PRINT REQUESTED
 +00080
 +00160
 +00240
 +00320
 +00400

5 508 PRINT REQUESTED
 +00080
 +00160

4 DATA_SET_NAME: PIONEER 11 TRAJECTORY EPHEMERIS ARCHIVE (Extension to
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K the equations of motion for a full set of acceleration
 models. For the Pioneer 11 trajectory ephemeris, DPTRAJ
 integration was limited to the N-body gravitation (planets, Sun and Moon) with Earth oblateness (J2),
 and a simple solar pressure model included.

8 Initial conditions for the DPTRAJ runs were obtained from various Orbit Determination (OD) solutions performed during the mission. A set of eleven spacecraft initial state vectors (position, velocity and epoch) was compiled to cover all phases of the mission from launch to 1 January 1990 (volume USA_NASA_NSSD_P11Z_0001).

(1) launch to first midcourse maneuver
 (2) First midcourse maneuver to second midcourse maneuver
 (3) Se

DC-033397
1/1/90 - 1/1/5000